Amendments to the Specification

Please amend the paragraph numbered 0011 in the as-filed application (0013 in the PGPub.) as follows:

~~The figure~~ FIG. 1 shows an exemplary combustor 20 positioned between compressor and turbine sections 22 and 24 of a gas turbine engine 26 having a central longitudinal axis or centerline 500. The exemplary combustor includes an annular combustion chamber 30 surrounding the centerline 500 and bounded by inner (inboard) and outer (outboard) walls 32 and 34 and a forward bulkhead 36 spanning between the walls. The bulkhead carries a circumferential array of swirlers 40 and associated fuel injectors 42. The exemplary fuel injectors extend through the engine diffuser case 44 to convey fuel from an external source to the associated injector outlet 46 at the associated swirler 40. The swirler outlet 48 thus serves as a principal fuel/air inlet to the combustor. One or more sparkplugs 50 are positioned with their working ends 52 along an upstream portion 54 of the combustion chamber 30 to initiate combustion of the fuel/air mixture. The combusting mixture is driven downstream within the combustor along a principal flowpath 504 through a downstream portion 56 to a combustor outlet 60 immediately ahead of a turbine fixed vane stage 62.

PAGE 4/10 * RCVD AT 10/18/2005 4:16:31 PM [Eastern Daylight Time] * SVR:USPTO-EFXRF-6/28 * DNIS:2738300 * CSID:203 865 0297 * DURATION (mm-ss):01-54